

ENGINE REPAIR MAINTENANCES RELEASES

AUG 2019

N-30584

ENGINE MAJOR OVERHAUL DATED 4-15-93

ENGINE MAINTENANCE RELEASES AND INVOICES.
OVERHAULED BY DAVID P. BOROWO, AWP 380648974IA.

LYCOMING O-360 A2F

SERIAL # L-13785-36A

United States of America

Department of Transportation—Federal Aviation Administration

Supplemental Type Certificate

Number SE1779NM

This certificate, issued to

Precise Flight, Inc.
63354 Powell Butte Road
Bend, OR 97701

*certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part * of the * Regulations.*

Original Product—Type Certificate Number:

*See attached FAA Approved Model List (AML)

Make:

No. SE1779NM for list of approved airplane models

Model:

and applicable airworthiness regulations

Description of the Type Design Change: Installation of Precise Flight, Inc. Standby Vacuum System SVS III, in accordance with Engineering Drawing 000V0000, Revision -, dated March 10, 2000, and Installation Report No. 50050, Revision 25, dated August 26, 1999, or later approved revision. Or Installation of Precise Flight, Inc. SVS V, in accordance with Precise Flight, Inc. Engineering Drawing 000V0000, Revision -, dated March 10, 2000 and Installation Instructions 08072, Revision -, dated December 12, 1999, or later FAA approved revision.

Limitations and Conditions: Approval of this change in type design applies to the above model aircraft only. This approval should not be extended to other aircraft of these models on which other previously approved modifications are incorporated unless it is determined by the installer that the relationship between this change and any of those other previously approved modifications, including changes in type design, will introduce no adverse effect upon the airworthiness of that aircraft. A copy of this Certificate, Continuation Sheet, and FAA approved AML No. SE1779NM, dated December 30, 1983, or later FAA approved revision, must be maintained as part of the permanent records for the modified aircraft.

United States Of America
Department of Transportation - Federal Aviation Administration

Supplemental Type Certificate

Number SA2586NM

This Certificate issued to J. P. INSTRUMENTS
PO Box 7033
Huntington Beach, CA 92646

certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part 3 of the Civil Aviation Regulations, including respective Amendments as specified in the attached Approved Model List.

Original Product Type Certificate Number: * See attached FAA Approved J.P. Instruments
Make: * Master Eligibility List No. SA2586NM for list
Model: * of approved aircraft models and applicable TCDS

Description of Type Design Change:

Installation of J. P. Instruments temperature monitoring systems in accordance with FAA Approved J. P. Instruments Drawing List Report No. 100, Revision D, dated December 19, 1996, or later FAA approved revisions. FAA Approved Airplane/Rotorcraft Flight Manual Supplement No. 1 for EGT-701 temperature indicator, Revision A, dated December 13, 1996, or later FAA approved revisions.

Limitations and Conditions: The approval of the change in type design applies to the basic airplane of the specific models that are otherwise unmodified. This approval should not be extended to other specific airplanes of these models on which other previously approved modifications are incorporated, unless it is determined that the interrelationship between this installation and any previously approved configuration will not introduce any adverse effect upon the airworthiness of that airplane. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission. (See continuation sheet)

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

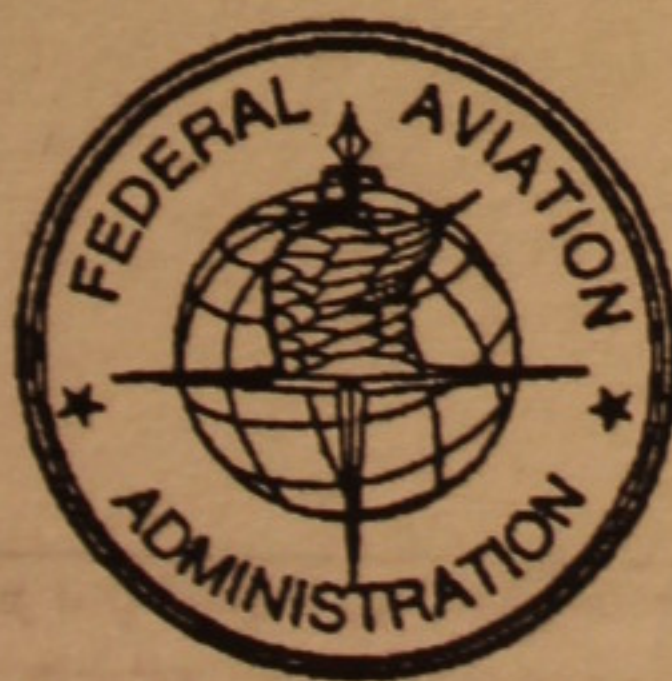
Date of application: December 31, 1984

Date of issuance: August 14, 1985

Date reissued:

Date amended: July 13, 1987, November 13, 1992, December 19, 1996, May 15, 1998, June 17, 1999

By direction of the Administrator



[Signature]
(Signature)
Manager, Propulsion Branch
Los Angeles Aircraft Certification Office

(Title)

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.

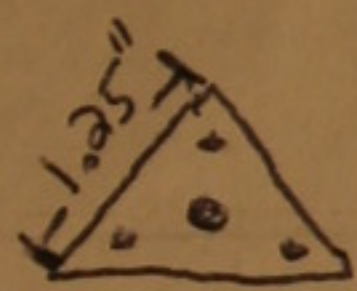
NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

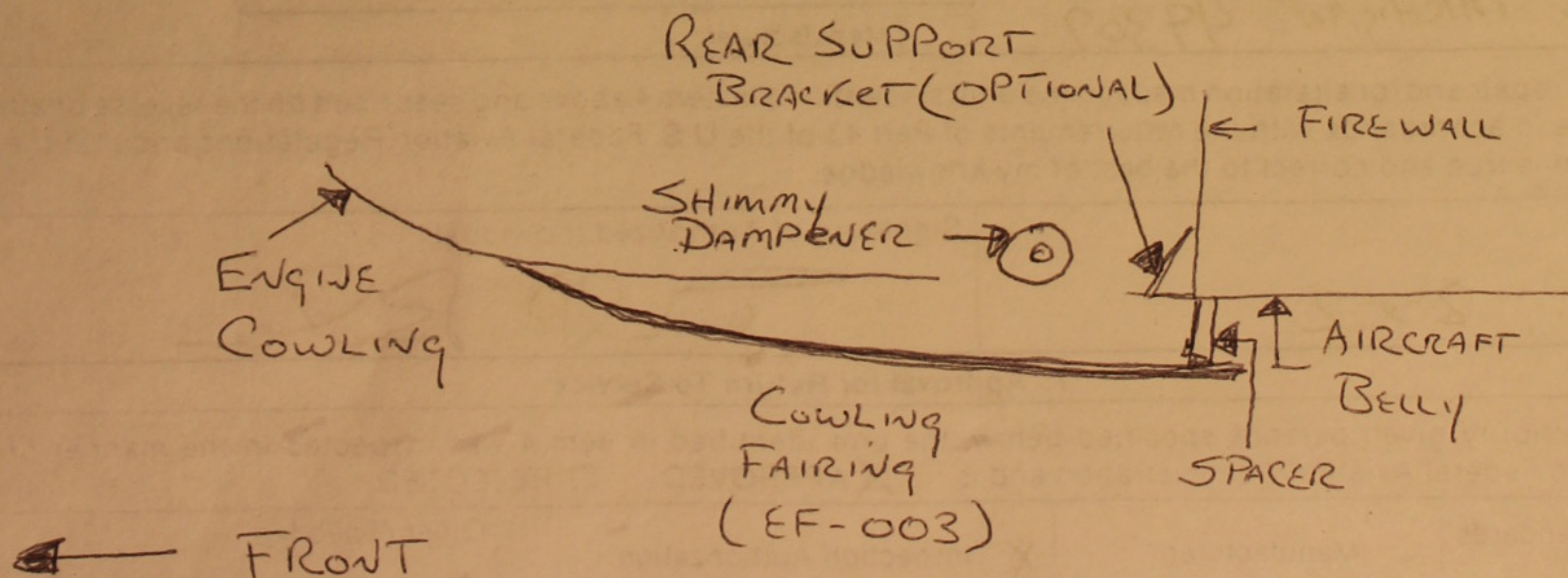
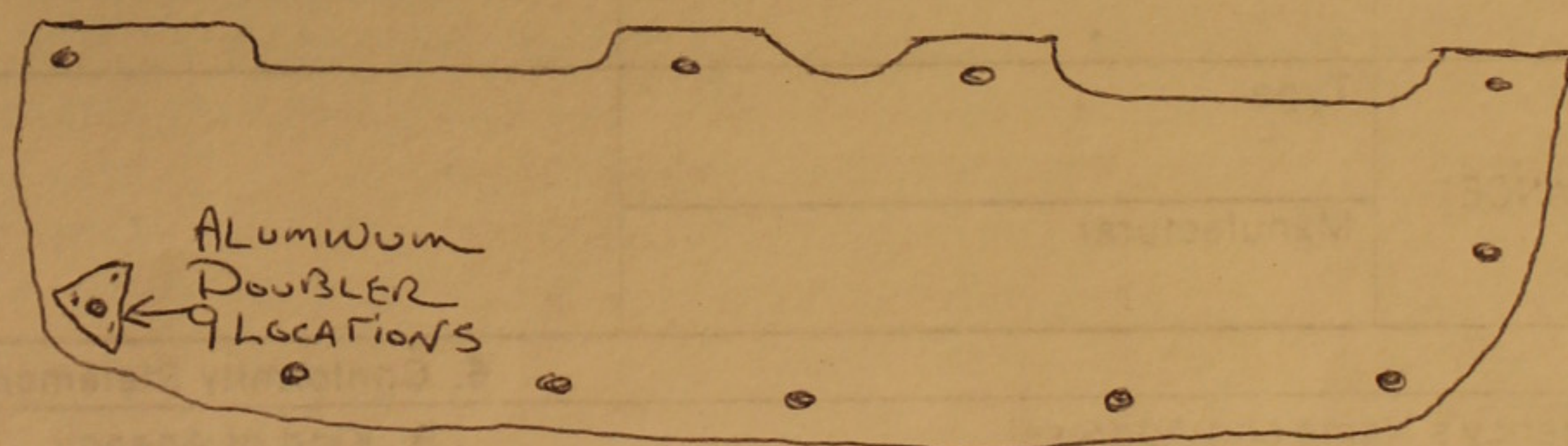
8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

ADDED LOWER COWLING FAIRING FABRICATED FROM FIBERGLASS AND POLYESTER RESIN. ATTACHED FAIRING TO BOTTOM COWLING USING 8-32 CS SCREWS AND MS 21059 SELF LOCKING NUT PLATES. ADDED ALUMINUM REINFORCING PLATES AT 9 LOCATIONS ON ADDED FAIRING. ALL WORK ACCOMPLISHED IN ACCORDANCE WITH 43.13-2A CH 1 PAR 4-10 AND SUGGESTED INSTALLATION PROCEDURES SUPPLIED BY MAPLE LEAF AVIATION LTD, PO BOX 160, BRANDON MANITOBA R7A-5Y8 ← END →



DRILL TO CLEAR
8-32 MACHINE SCREWS
#40 DRILL (3 PLACES)



☐ Additional Sheets Are Attached

INSTRUCTIONS FOR CONTINUED AIRWORTHINESS

A/C Make: CESSNA. Model: 177A S/N: 17701344 ~~17701344~~ DB Reg. N30584
#:

Revision: Date: MARCH, 1, 2002

This sixteen item checklist are Instructions for Continued Airworthiness (ICA), to comply with FAA Handbook Bulletin for Airworthiness (HBAW 98-18 Dated October 7, 1998), are applicable to the aircraft listed above when the following equipment is installed:

SYSTEM:

ITEM	CHECKLIST INFORMATION
1.	Introduction: This section briefly describes the aircraft, engine, propeller, or component that has been altered. Include any other information on the content, scope, purpose, arrangement, applicability, definitions, abbreviations, precautions, units of measurement, referenced publications, and distribution of the ICA as applicable. Comment: INSTALLATION OF A FIBERGLASS FAIRING WHICH FITS ON THE LOWER ENGINE COWLING. THE FAIRING COVERS A LARGE OPENING IN THE AREA OF THE SHIMMY DAMPENER THE PURPOSE OF THE FAIRING IS TO INCREASE THE SPEED OF THE AIRCRAFT.
2.	Description: Of the major alteration, its functions, including an explanation of its interface with other systems, if any. Comment: INSTALLATION OF A LOWER COWLING FAIRING (EF-003) MADE FROM FIBERGLASS AND IS ATTACHED BY 9(NINE) # 8-32 SCREWS AND NUT PLATES RIVETED TO THE ENGINE COWLING.
3.	Control: Operation information: Or special procedures, if any. Comment: NONE
4.	Servicing information: Such as types of fluids used, servicing points, and location of access panels, as appropriate. Comment: THE FIBERGLASS FAIRING (EF-003) WOULD HAVE TO BE REMOVED TO SERVICE THE SHIMMY DAMPENER AND TO REMOVE THE ENGINE COWLING.
5.	Maintenance Instructions: Such as recommended inspection/maintenance periods in which each of the major alteration components are inspected, cleaned, lubricated, adjusted, tested, including applicable wear tolerances and work recommended at each scheduled maintenance period. This section can refer to the manufacturers' instructions for the equipment installed where appropriate (e.g., functional checks, repairs, inspections.) It should also include any special notes, cautions, or warnings, as applicable. Comment: CHECK SECURITY AND CONDITION OF FAIRING PRIOR TO EACH FLIGHT.

6.

Trouble shooting information: Information describing probable malfunctions, how to

recognize those malfunctions, and the remedial actions to be taken. **Comment:**

THE INSTALLED FIBERGLASS FAIRING (EF-003) COVERS AN AREA OF THE LOWER ENGINE COWLING WHICH PROVIDES AN EXIT FOR THE AIR THAT PROVIDES ENGINE COOLING, AND REDIRECTS THIS AIR TO ESCAPE THROUGH SLOTS FORMED IN THE REAR EXITS OF THE FIBERGLASS FAIRING (EF-003)

ON THE FIRST FLIGHT AFTER INSTALLATION OF THE FIBERGLASS FAIRING - (EF-003) ENGINE OIL TEMPS AND CHT TEMPS ARE TO BE OBSERVED AND IF NOTED TO BE HIGH, THE FORMED REAR AIR EXITS OF THE FIBERGLASS FAIRING (EF-003) MAY BE TRIMMED TO INCREASE THE AIR FLOW THROUGH THE AIR EXITS, INCREASING ENGINE COOLING.

7. **Removal and replacement information:** This section describes the order and method of removing and replacing products, parts and any necessary precautions. This section should also describe or refer to manufacturer's instructions to make required tests, trim checks, alignment, calibrations, center of gravity changes, lifting or shoring, etc., if any. **Comment:**

SEE # 6

8. **Diagrams:** Of access plates and information, if needed, to gain access for inspection.

Comment:

NONE

9. **Special inspection requirements:** Such as X-ray, ultrasonic testing, or magnetic particle inspection, if required. **Comment:**

DOES NOT APPLY.

10. **Application of protective treatments:** To the affected area after inspection and/or maintenance, if any. **Comment:**

DOES NOT APPLY.

11. **Data:** Relative to structural fasteners such as type, torque, and installation requirements, if any. **Comment:**

THE FIBERGLASS FAIRING (EF-003) IS ATTACHED TO THE LOWER ENGINE COWL BY 9 NINE #8-32 MACHINE SCREWS AND 9 NINE SELF LOCKING NUT PLATES MS 21059.

12. **List of special tools:** Special tools that are required, if any. **Comment:**

NONE.

13. **For commuter category aircraft:** The following additional information must be furnished, as applicable: A. Electrical loads B. Methods of balancing flight controls C. Identification of primary and secondary structures D. Special repair methods applicable to the airplane. **Comment:**

DOES NOT APPLY.

14. **Recommended overhaul periods:** Are required to be noted on the ICA when an overhaul

period has been set by the manufacturer of a component, or equipment. If there is no overhaul period, the ICA should state for item 14: "No additional overhaul time limitations." **Comment:**

"NO ADDITIONAL OVERHAUL TIME LIMITATIONS"

15.	Airworthiness Limitation Section: Include any "approved" airworthiness limitations identified by the manufacturer or FAA Type Certificate Holding Office (e.g., An STC incorporated in a larger field approved major alteration may have an airworthiness limitation.) The FAA inspector should not establish, alter, or cancel airworthiness limitations without coordinating with the appropriate FAA Type Certificate Holding Office. If there are no changes to the airworthiness limitations, the ICA should state for item 15: "No additional airworthiness limitations" or "Not Applicable." Comment: APPROVED FOR CESSNA 177 SERIAL # 000001 THRU 01370 PER CANADA LIMITED SUPPLEMENTAL TYPE APPROVAL NUMBER C-LSA 94-242
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16.	Revision: This section should include information on how to revise the ICA. For example, a letter will be submitted to the local FSDO with a copy of the revised FAA Form 337 and revised ICA. The FAA inspector accepts the change by signing Block 3 and including the following statement: "The attached revised/new Instructions for Continued Airworthiness (date_____) for the above aircraft or component major alteration have been accepted by the FAA, superseding the Instructions for Continued Airworthiness (date_____)." Once the revision has been accepted, a maintenance record entry will be made, identifying the revision, its location, date of the Form 337. Comment:
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Note:
Implementation and Record Keeping: For major alterations performed in accordance with FAA Field Approval policy, the owner/operator operating under part 91 is responsible for ensuring that the ICA is made part of the applicable section 91.409 inspection program for their aircraft. This is accomplished when a maintenance entry is made in the aircraft's maintenance record in accordance with section 43.9. This entry records the major alteration and identifies the original ICA location (e.g., Block 8 of FAA Form 337, dated 5/28/98) along with a statement that the ICA is now part of the aircraft's inspection/maintenance requirements.

For major alterations performed in accordance with a field approval on air carrier aircraft, the air carrier operator is responsible for ensuring that the ICA is made part of the applicable inspection/maintenance program for their aircraft. If a procedure is not currently included in the operator's manual to incorporate ICA, this process will need to be appropriately addressed (i.e. the operator submits a revision to its maintenance program to the applicable certificate-holding district office (CHDO)).

For aircraft inspected under an Approved Aircraft Inspection Program (AAIP), the operator will submit a change to the CHDO in accordance with section 135.419 b).

For air carrier aircraft inspected using an annual/100 hour inspection program, a reference to the new ICA will be made in the aircraft's maintenance record in accordance with section 43.9. This entry records the major alteration and identifies the original ICA location (e.g., ICA are located/attached to Block 8 of FAA Form 337, dated 5/28/98). In addition, the operator will request a revision to the operator's Operations Specifications, additional maintenance requirements, which incorporates the ICA into the inspection program.



U.S. Department
of Transportation
Federal Aviation
Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only

Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make CESSNA	Model 177-A
	Serial No. 177 01344	Nationality and Registration Mark N 30584
2. Owner	Name (As shown on registration certificate) PETER VANDE MARK	Address (As shown on registration certificate) 17349 160TH AVE ROONEY MICHIGAN.

3. For FAA Use Only

The data identified herein complies with the applicable airworthiness requirements and is approved for the above described aircraft, subject to conformity inspection by a person authorized in FAR Part 43, Section 43.7.

CRR FSDO
District Office

04/11/02
Date

Hemphill
Signature of FAA Inspector

4. Unit Identification

5. Type

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	~~~~~ (As described in Item 1 above) ~~~~~				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

RECEIVED
APR 16 2002
BY:

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
DAVID P. BORDANO 21158 WEST DR. BIG RAPIDS, MICHIGAN. 49307	<input checked="" type="checkbox"/> U.S. Certificated Mechanic	A&P 38064 8976
	<input type="checkbox"/> Foreign Certificated Mechanic	
	<input type="checkbox"/> Certificated Repair Station	
	<input type="checkbox"/> Manufacturer	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date MARCH 1, 2002	Signature of Authorized Individual D.P. Bordano
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7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ APPROVED ☐ REJECTED

BY	FAA Fit. Standards Inspector	Manufacturer	<input checked="" type="checkbox"/> Inspection Authorization	Other (Specify)
	FAA Designee	Repair Station	Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection 4-12-2002		Certificate or Designation No. 38064 8976 IA	Signature of Authorized Individual D.P. Bordano	



US Department
of Transportation
Federal Aviation
Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only

Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make CESSNA	Model 177-A
	Serial No. 177 01344	Nationality and Registration Mark N38584
2. Owner	Name (As shown on registration certificate) PETER VANDE MARK	Address (As shown on registration certificate) 17349 160TH AVE RODNEY, MICHIGAN.

3. For FAA Use Only

4. Unit Identification				5. Type	
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	(As described in Item 1 above)				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
DAVID P. BORDANO 21158 WEST DRIVE BIG RAPIDS, MICHIGAN. 49307	<input checked="" type="checkbox"/> U.S. Certificated Mechanic	A+P 380648976
	<input type="checkbox"/> Foreign Certificated Mechanic	
	<input type="checkbox"/> Certificated Repair Station	
	<input type="checkbox"/> Manufacturer	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date 1-23-2001	Signature of Authorized Individual
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7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ APPROVED ☐ REJECTED

BY	FAA Fit. Standards Inspector	Manufacturer	<input checked="" type="checkbox"/>	Inspection Authorization	Other (Specify)
	FAA Designee	Repair Station		Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection 1-23-2001		Certificate or Designation No. 380648976IA.		Signature of Authorized Individual 	

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.


8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

INSTALLED J.P.I. ENGINE SCANNER AND EGT/CHT
THERMOCOUPLES IN ACCORDANCE WITH STC #SA 2586 NM
INSTALLED IN ACCORDANCE WITH INSTALLATION MANUAL #103
REVISION C, PRINTED 6-7-2000.

← END →

☐ Additional Sheets Are Attached

 U.S. Department of Transportation Federal Aviation Administration	MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)	Form Approved OMB No. 2120-0020
		For FAA Use Only
		Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

Aircraft	Make CESSNA	Model 177-A
	Serial No. 177 01344	Nationality and Registration Mark N30584
Owner	Name (As shown on registration certificate) PETER VANDE MARK	Address (As shown on registration certificate) 17349 160TH AVE RODNEY, MICHIGAN

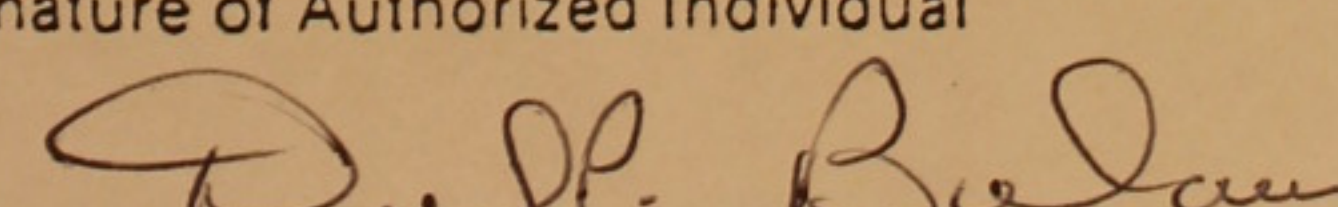
3. For FAA Use Only

4. Unit Identification				5. Type	
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	~~~~~ (As described in Item 1 above) ~~~~~				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

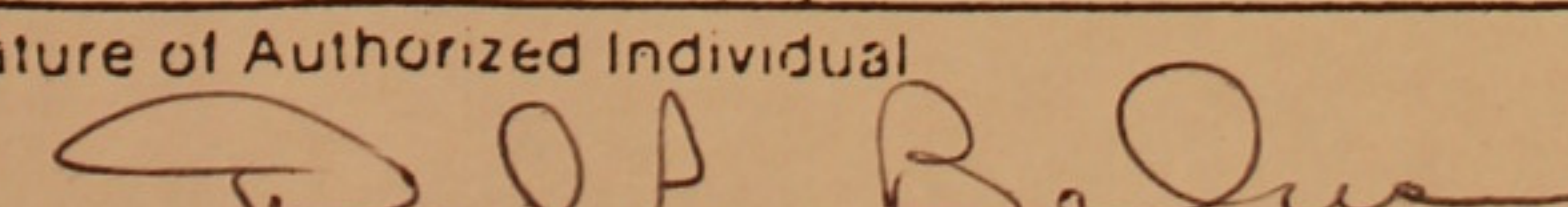
A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
DAVID P. BORDANO 21158 WEST DRIVE BIG RAPIDS, MICHIGAN 49307	<input checked="" type="checkbox"/> U.S. Certificated Mechanic	A-P 380648976
	<input type="checkbox"/> Foreign Certificated Mechanic	
	<input type="checkbox"/> Certificated Repair Station	
	<input type="checkbox"/> Manufacturer	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date 4-15-2004	Signature of Authorized Individual 
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7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ APPROVED ☐ REJECTED

BY	FAA Flt. Standards Inspector	Manufacturer	<input checked="" type="checkbox"/> Inspection Authorization	Other (Specify)
	FAA Designee	Repair Station	Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection 4-15-2004		Certificate or Designation No. A-P 380648976 IA	Signature of Authorized Individual 	

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

INSTALLATION OF PRECISE FLIGHT INE. STAND-BY VACUUM
SYSTEM IN ACCORDANCE WITH STC # SA 2126 NM AND
PRECISE INSTALLATION GUIDE NUMBER 08072 DATED 12-2-99
← END →

JONESM

TERMS		REFERENCE NUMBER	SHIP VIA	ORDER NO.	ORDERED	REQUESTED	SHIPPED	CUSTOMER REPRESENTATIVE					
Cash/Code NO.	ITEM NUMBER	ITEM DESCRIPTION		ITEM REFERENCE	O.	A X	QUANTITY PER	PRICE PER	Dennis Wyman		Thomas A. Hall		
									Quantity	List/UOM	Discount	Sell/UOM	Extended
07===== G & N AIRCRAFT, INC. =====													
INSPECT & ADVISE 4) 360 CYLINDERS WITH VALVES & PISTONS													
LABOR TO RECONDITION 2) CYLINDERS CLEAN PISTONS & FIT RINGS													
SOLD 2) EXCHANGE CYLINDERS													
Labor Totals													
Parts Section:													
CL12C		Class 12 Chrome Cyl Stud Assbl				Each	2.00	440.00	NET		440.00		880.00
AEL19001		Valve (SL19001A)				Each	1.00	122.50	5.9%		115.24		115.25
74673		Ring (SL74673)				Each	4.00	33.60	20.0%		26.88		107.52
73998		Ring (SL73998)				Each	2.00	16.40	20.0%		13.12		26.24
74241		Ring				Each	4.00	34.53	20.0%		27.62		110.50
14H21950		Rings				Each	2.00	26.85	20.0%		21.48		42.96
SL12032-SC		Gasket kit				Each	4.00	8.40	10.5%		7.51		30.06
72198		Plug				Each	8.00	28.52	NET		28.52		228.22
17C21191		Key				Each	8.00	5.75	20.0%		4.60		36.80
60009		Key				Each	8.00	1.05	17.6%		.86		6.92
AEL74230P20		Guide				Each	2.00	12.32	20.0%		9.85		19.71
AEL74230P05		Guides				Each	2.00	12.32	NET		.00		.00
31C12		Stud				Each	4.00	3.30	NET		.00		.00
Parts Totals													
Freight Section:													
Continued on Next Page.													

G&N Aircraft, Inc.

Phone: 219-924-7110

(800) 348-6504

Fax: (219) 924-1059

WORK ORDER INVOICE

WORK ORDER: 024/57717

FAA Certified Repair Station MW2R020L

PLEASE MAKE CHECKS PAYABLE TO: G & N AIRCRAFT, INC.

4) 360 CYLINDERS

P/N: . S/N: .

B 2740
I Bordano Air Inc.
L 21158 West Dr.
L Big Rapids MI 49307-000

S 2740
H Bordano Air Inc.
I 21158 West Dr.
P Big Rapids MI 49307-000

JONESM

TERMS		REFERENCE NUMBER	SHIP VIA	ORDER NO.	ORDERED	REQUESTED	SHIPPED	CUSTOMER REPRESENTATIVE			
Cash/CC	ITEM NUMBER	ITEM DESCRIPTION	ITEM REFERENCE	QUANTITY PER	PRICE PER	Dennis Wyman		Thomas A. Hall			
						Quantity	List/UOM	Discount	Sell/UOM	Extended	
UPS CHARGE										59.13	
Freight Totals										59.13	
Operation 0007 Totals:						Flat Rated				2169.31	
						G & N AIRCRAFT, INC.			Total:	2,169.31	